



# SPIRE: Herschel's Submillimetre Camera and Spectrometer

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+ the SPIRE Consortium

- SCIENTIFIC GOALS AND DESIGN DRIVERS
- INSTRUMENT DESCRIPTION
- OBSERVING MODES
- PERFORMANCE ESTIMATES



# The SPIRE Consortium

Canada



China



France



Italy



Spain



Sweden



UK



USA



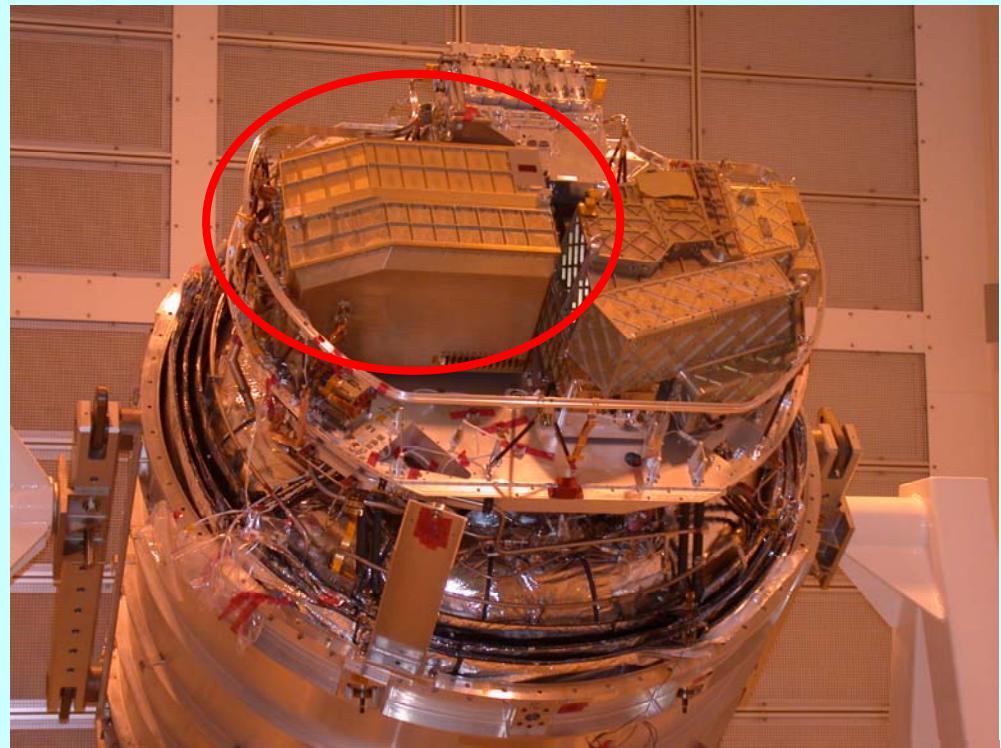
- Cardiff University, UK
- CEA Service d'Astrophysique, Saclay, France
- Institut d'Astrophysique Spatiale, Orsay, France
- Imperial College, London, UK
- Instituto de Astrofisica de Canarias, Tenerife, Spain
- Istituto di Fisica dello Spazio Interplanetario, Rome, Italy
- Jet Propulsion Laboratory/Caltech, Pasadena, USA
- Laboratoire d'Astronomie Spatiale, Marseille, France
- Mullard Space Science Laboratory, Surrey, UK
- NAOC, Beijing, China
- Observatoire de Paris, Meudon, France
- Rutherford Appleton Laboratory, Oxfordshire, UK
- Stockholm Observatory, Sweden
- UK Astronomy Technology Centre, Edinburgh, UK
- Università di Padova, Italy
- University of Lethbridge, Canada

- **3-band imaging photometer**

- 250, 360, 520  $\mu\text{m}$   
**(simultaneous)**
- $\lambda/\Delta\lambda \sim 3$
- 4 x 8 arcminute field of view
- Diffraction limited beams  
**(17, 24, 35")**

- **Imaging Fourier Transform Spectrometer**

- 200 - 670  $\mu\text{m}$   
**(complete range covered simultaneously)**
- 2.6 arcminute field of view
- $\Delta\sigma = 0.04 \text{ cm}^{-1}$  ( $\lambda/\Delta\lambda \sim 20 - 1000$  at 250  $\mu\text{m}$ )





# Instrument Design Drivers

- **Photometer**
  - Efficient multi-band mapping with largest possible field of view
  - Point and compact source observation with high efficiency
- **Spectrometer**
  - Point source survey spectroscopy with broad wavelength range
  - Imaging spectroscopy with maximum available field of view
  - Variable spectral resolving power (few x 10 to ~ 1000)
- **Both**
  - Simplicity, affordability, reliability, ease of operation
  - Complementary to other Herschel instruments and other facilities

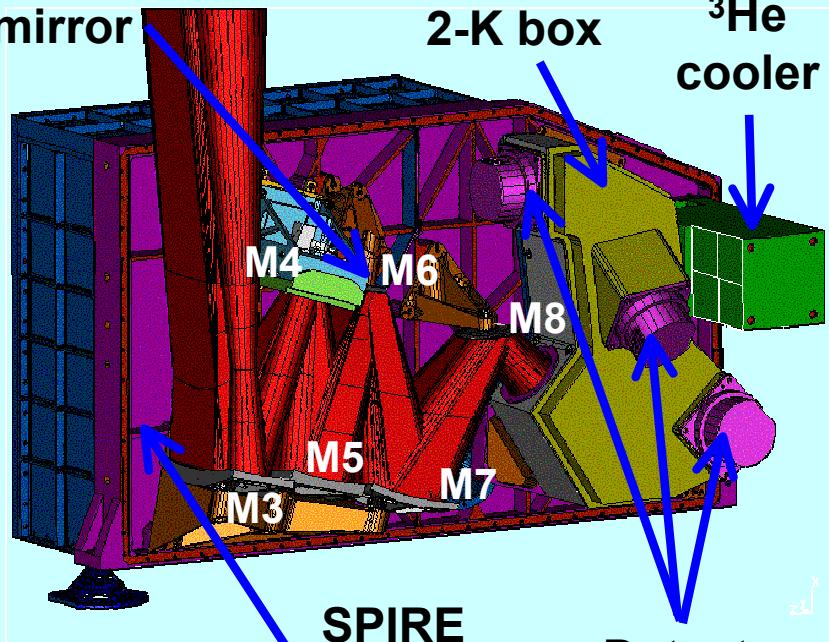


## SPIRE Scientific Goals: GT programmes as examples

- **High-redshift galaxies**
- **Local galaxies**
- **Star formation**
- **Interstellar medium**
- **Circumstellar matter**
- **Solar system**

# Photometer Layout and Optics

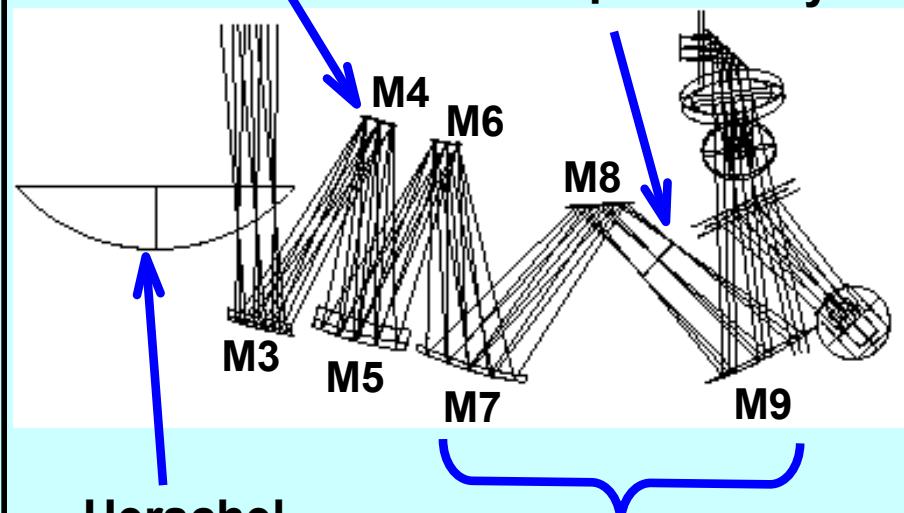
Beam steering mirror



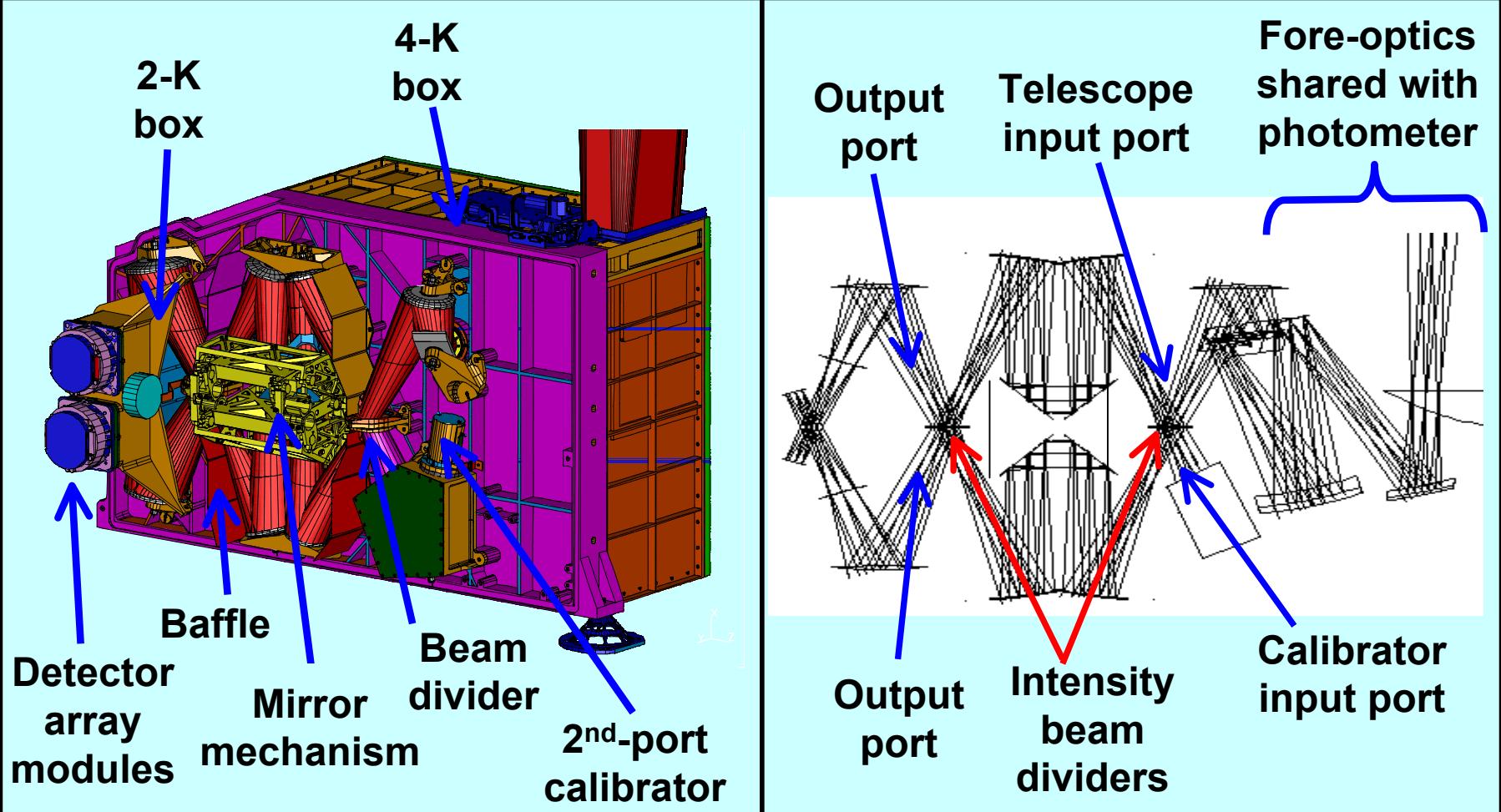
Beam steering mirror

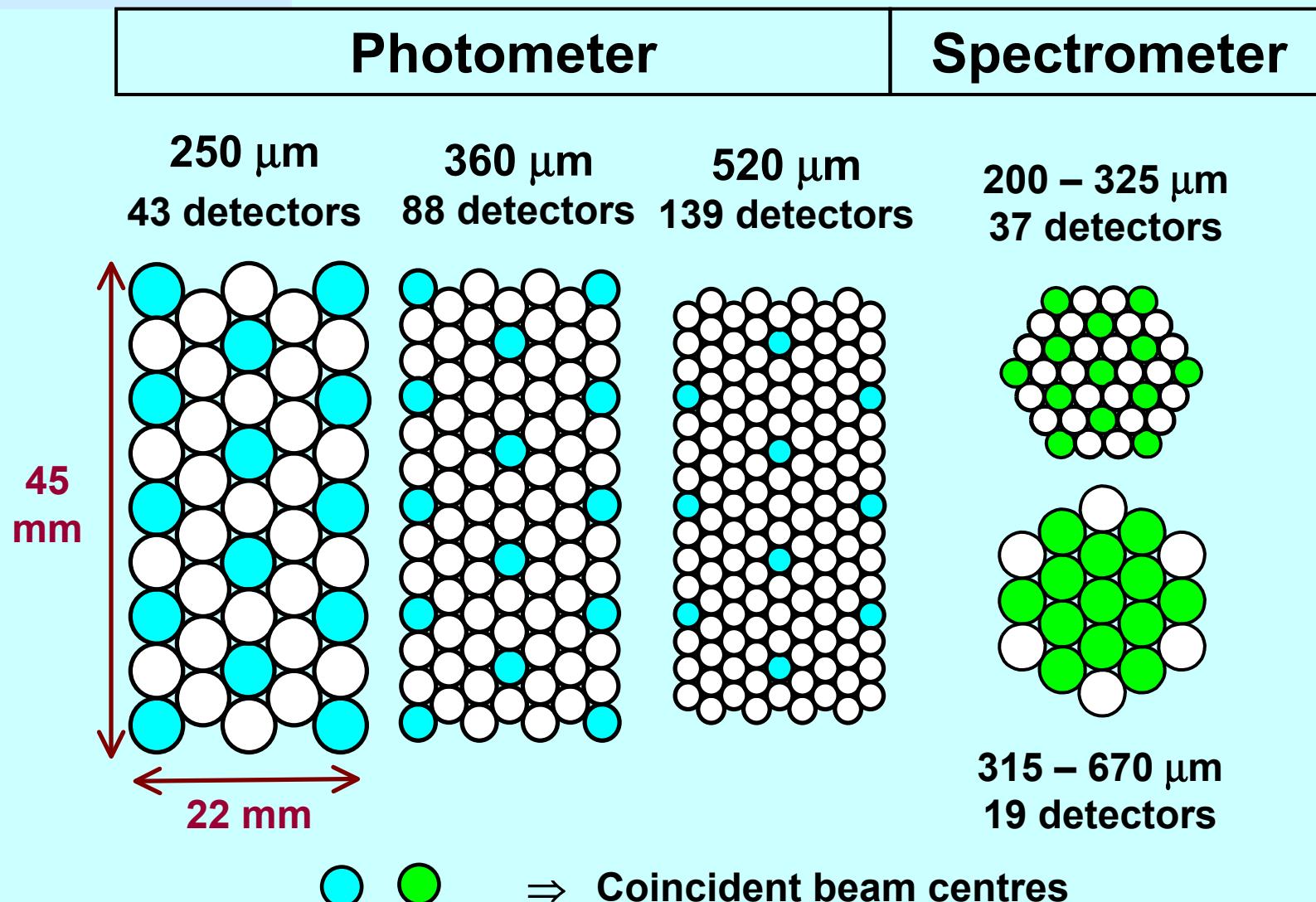
2-K cold stop

Dichroics and arrays



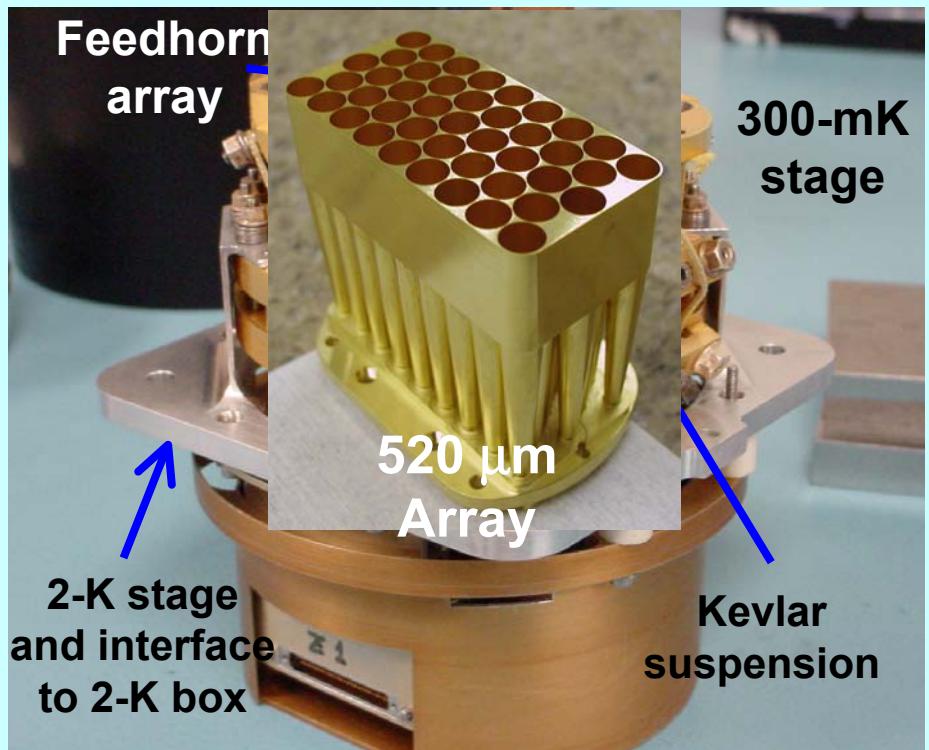
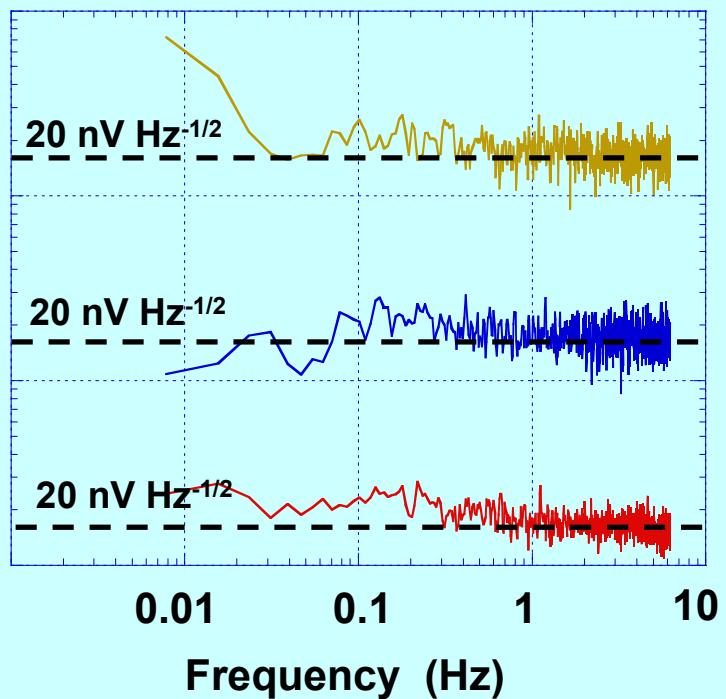
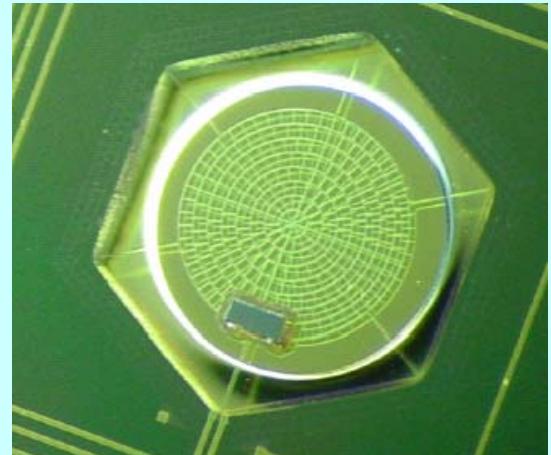
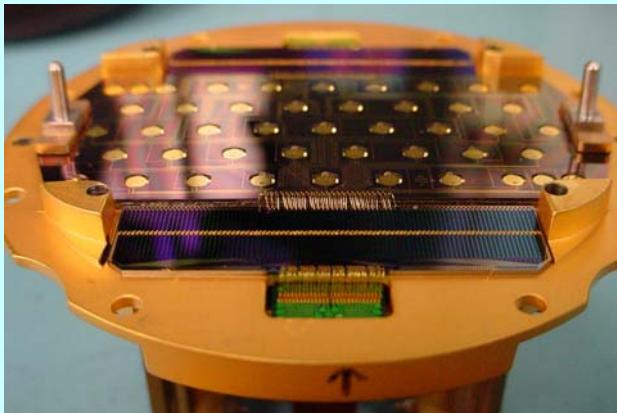
# FTS Layout and Optics





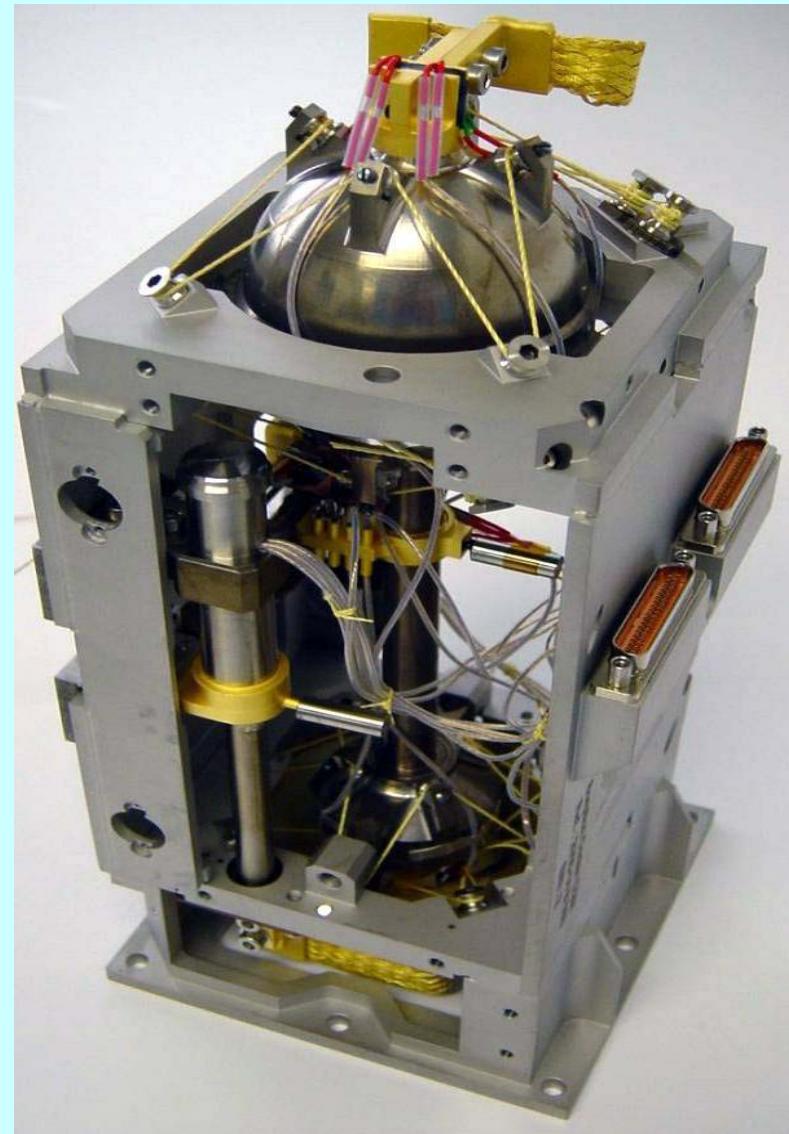
# 0.3-K Germanium Bolometer Arrays

- NEP  $\sim 3 \times 10^{-17}$  W Hz $^{-1/2}$
- 100-K Si JFET readout
- 1/f noise knee < 100 mHz

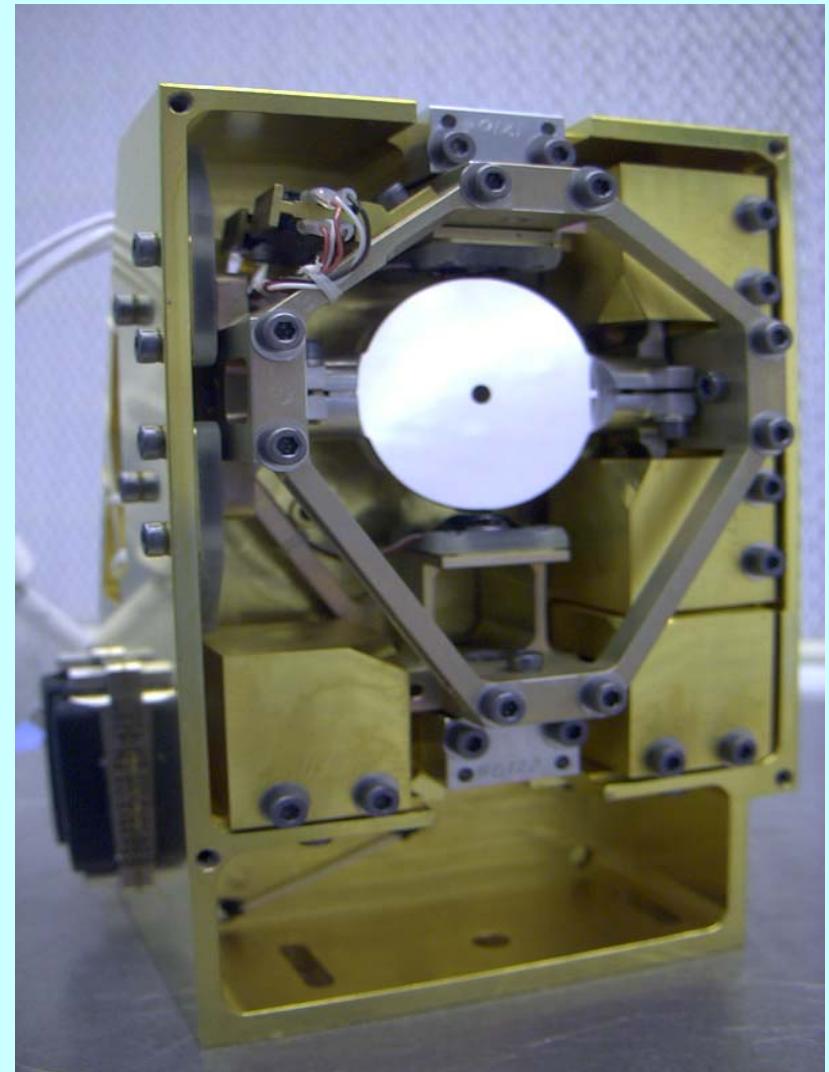
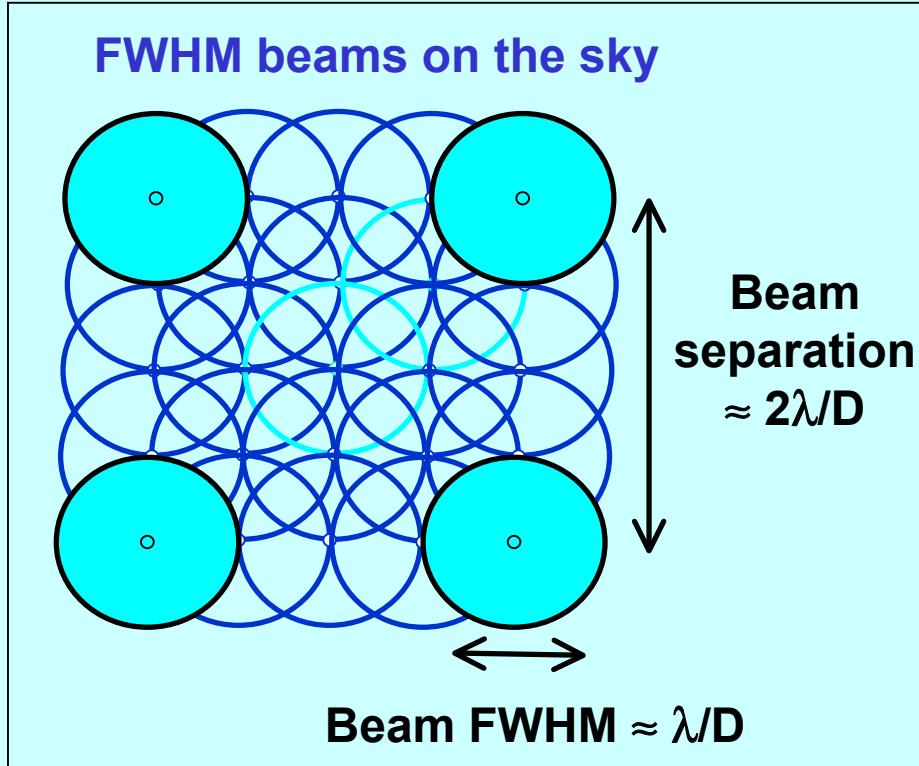


## <sup>3</sup>He Cooler

- Cold stage temp. < 290 mK
- Hold time > 46 hrs
- Cycle time < 2 hrs
- Gas-gap heat switches  
(no moving parts)
- Heat lift provided to  
detector arrays > 10  $\mu$ W

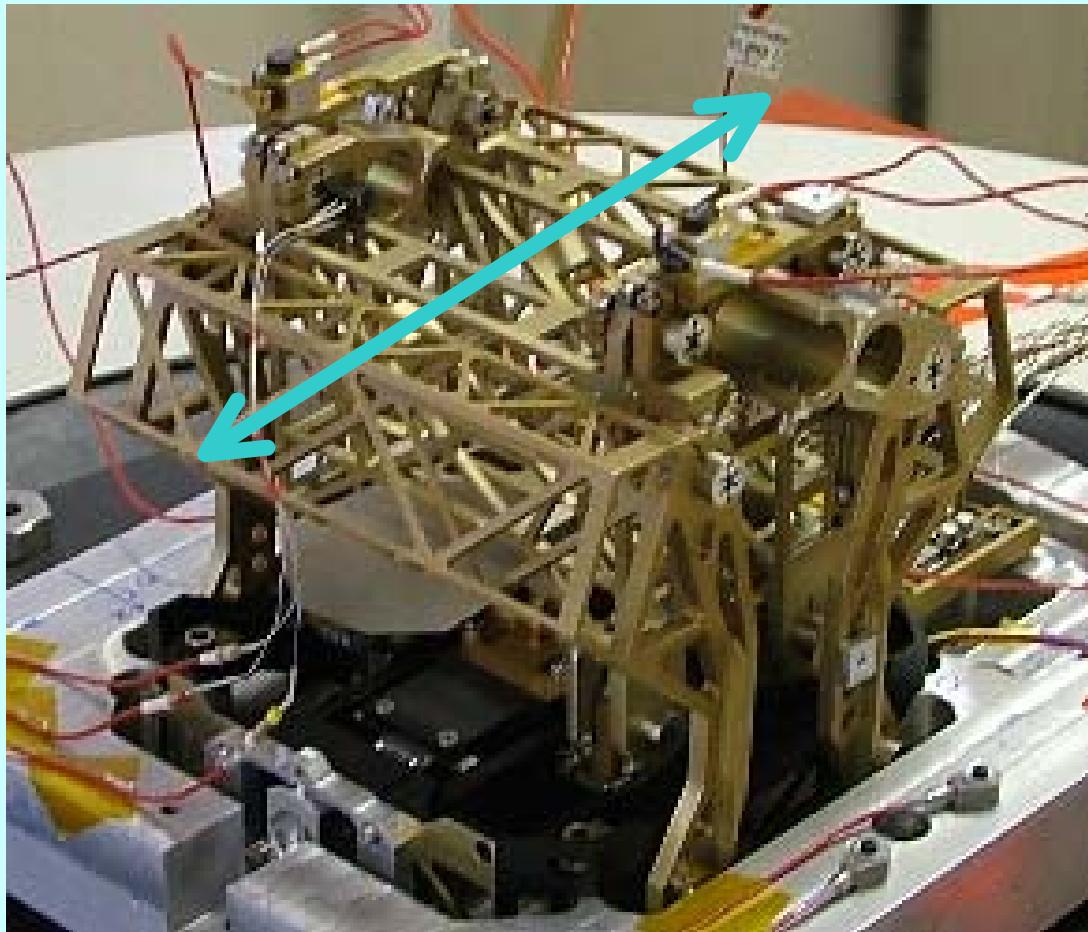


# Beam Steering Mechanism



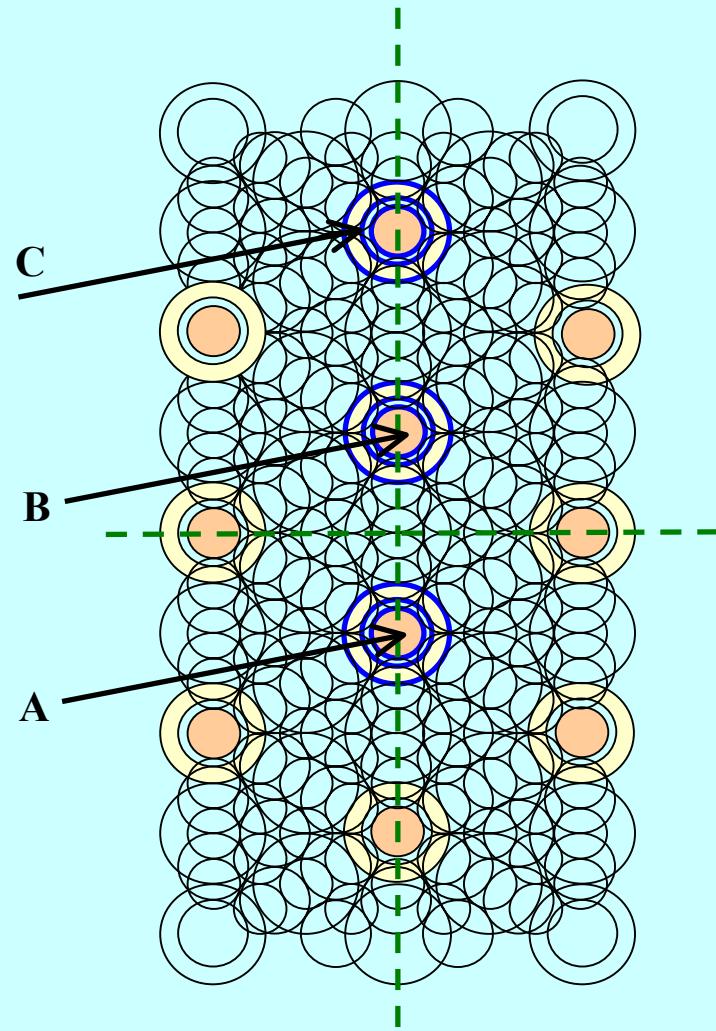
## FTS Scan Mechanism

- Double parallelogram carriage with toothless gear
- Moiré fringe position measurement system ( $0.1 \mu\text{m}$  accuracy)
- Continuous scan or step-and-integrate operation
- Nominal speed:  $0.5 \text{ mm s}^{-1}$ 
  - Signal frequency range 3 - 10 Hz
- 3.8-cm travel



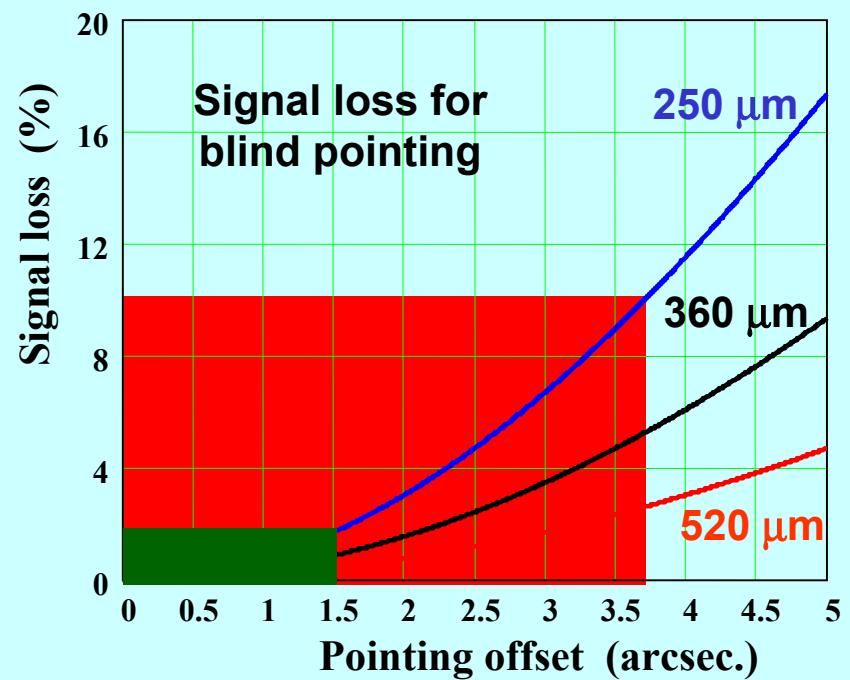
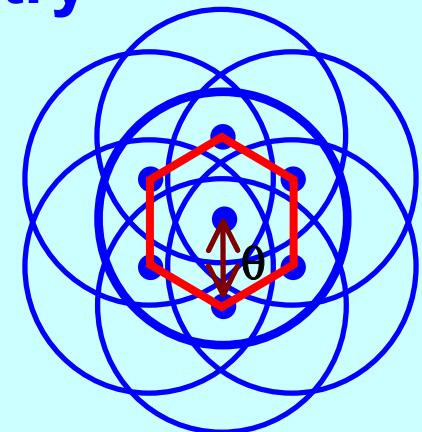
# Photometer Observing Modes: Point Source Photometry

- Simultaneous observation in the three bands with sets of co-aligned detectors
- Chopping 126" and nodding, using detector sets A, B, C
- OK if the pointing is accurate enough ( $\sim 1.5''$ )



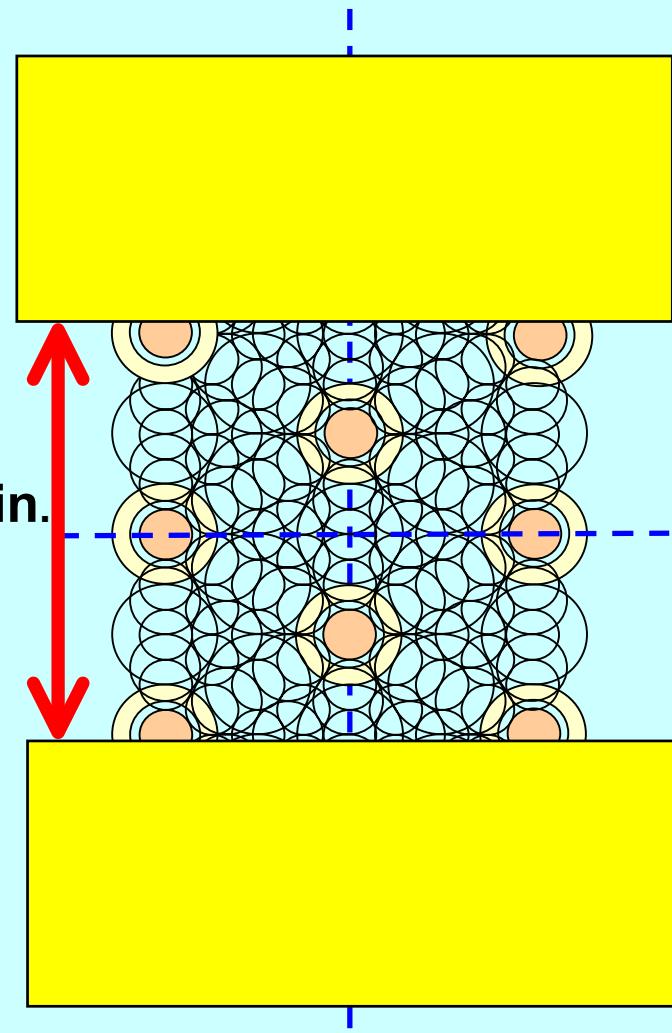
# Photometer Observing Modes: Seven-point Jiggle Photometry

- Chopping 126"
- 7-point map
- Angular step  $\theta \sim 5$  arcseconds (> pointing or positional error)
- Total flux and position fitted
- Compared to single accurately pointed observation, S/N for same total integration time is only degraded by
  - ~ 20% at 250  $\mu\text{m}$
  - ~ 13% at 360  $\mu\text{m}$
  - ~ 6% at 520  $\mu\text{m}$



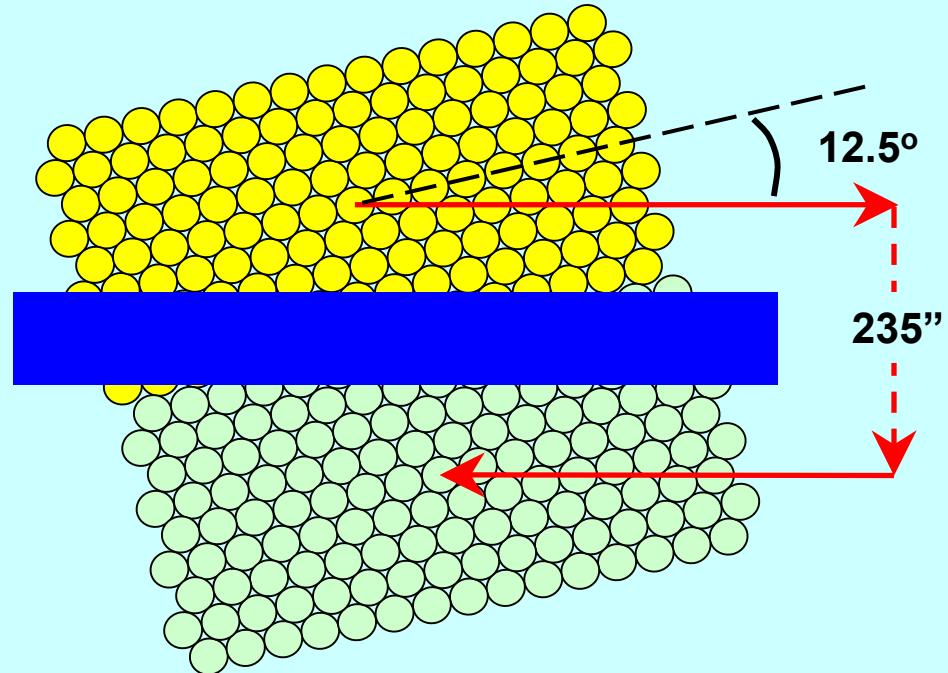
## Field (Jiggle) Map

- Telescope pointing fixed or in raster mode
- Chopping  $\pm 2$  arcmin
- 64-point “jiggle” pattern for full spatial sampling
- Available FoV =  $4 \times 4$  arcmin.



# Photometer Observing Modes: Scan Map

- Most efficient mode for large-area surveys
  - Telescope scans continuously at up to  $60''/\text{sec.}$
  - Scan parameters optimised for full spatial sampling and uniform distribution of integration time
  - Map of large area is built up from overlapping parallel scans
- Overlap region





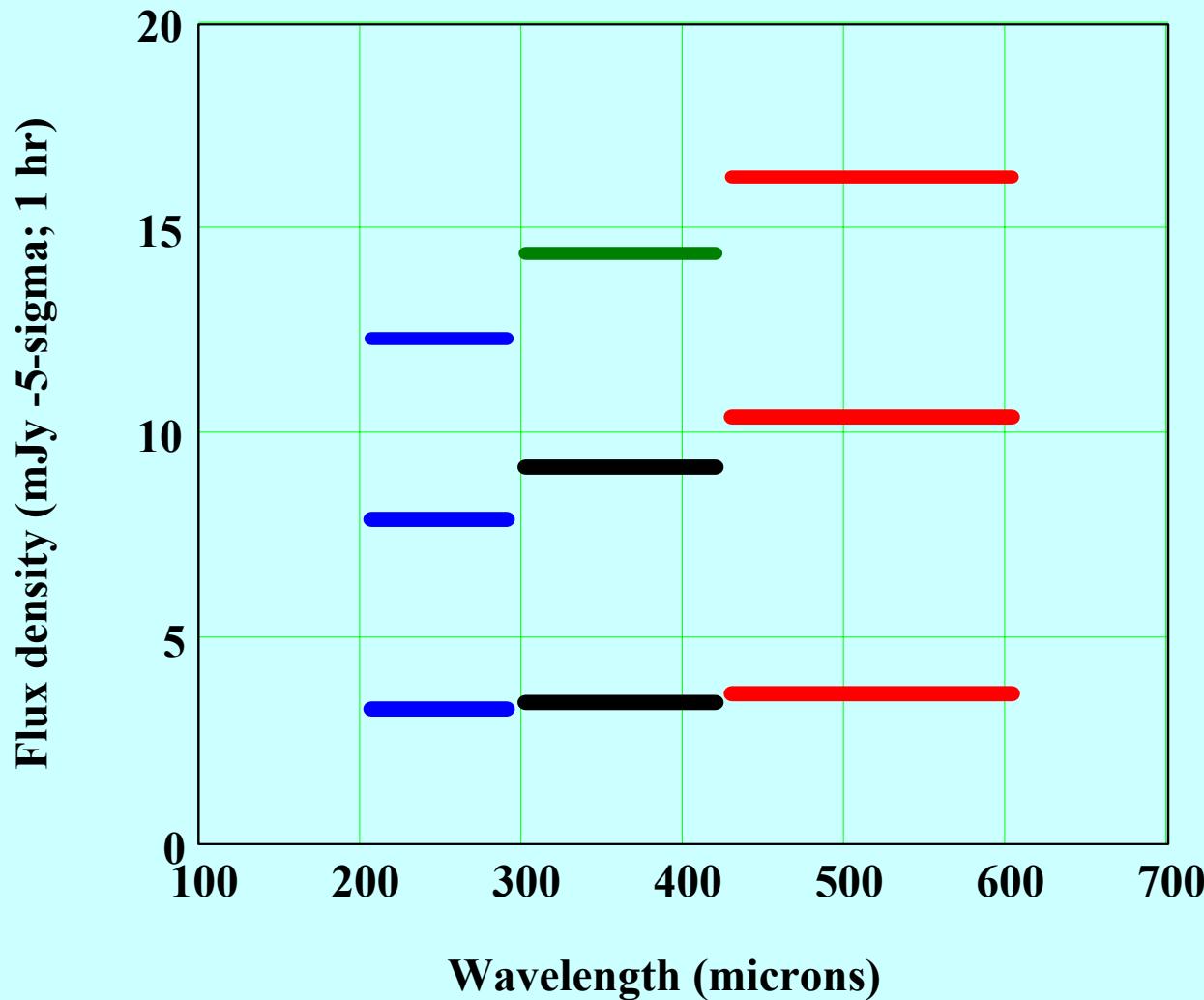
## FTS Observing Modes

- $\Delta\sigma = 0.04 - 2 \text{ cm}^{-1}$  by adjusting scan length
- **Continuous scan (nominal mode):**
  - Calibrator in 2<sup>nd</sup> port nulls telescope background
- **Step-and-integrate:**
  - 2<sup>nd</sup> port calibrator is off
  - Mirror stepped with integration at each position
  - Beam Steering Mechanism chops on the sky
- **Point source or sparse map spectroscopy/spectrophotometry**
  - Produces 2.6- arcmin sparse map – background is characterised by adjacent pixels
- **Imaging spectroscopy (fully-sampled)**
  - Beam steering mirror adjusts pointing between scans to acquire fully-sampled spectral image



# Sensitivity: Photometry

( $5\sigma$ ; 1 hr)

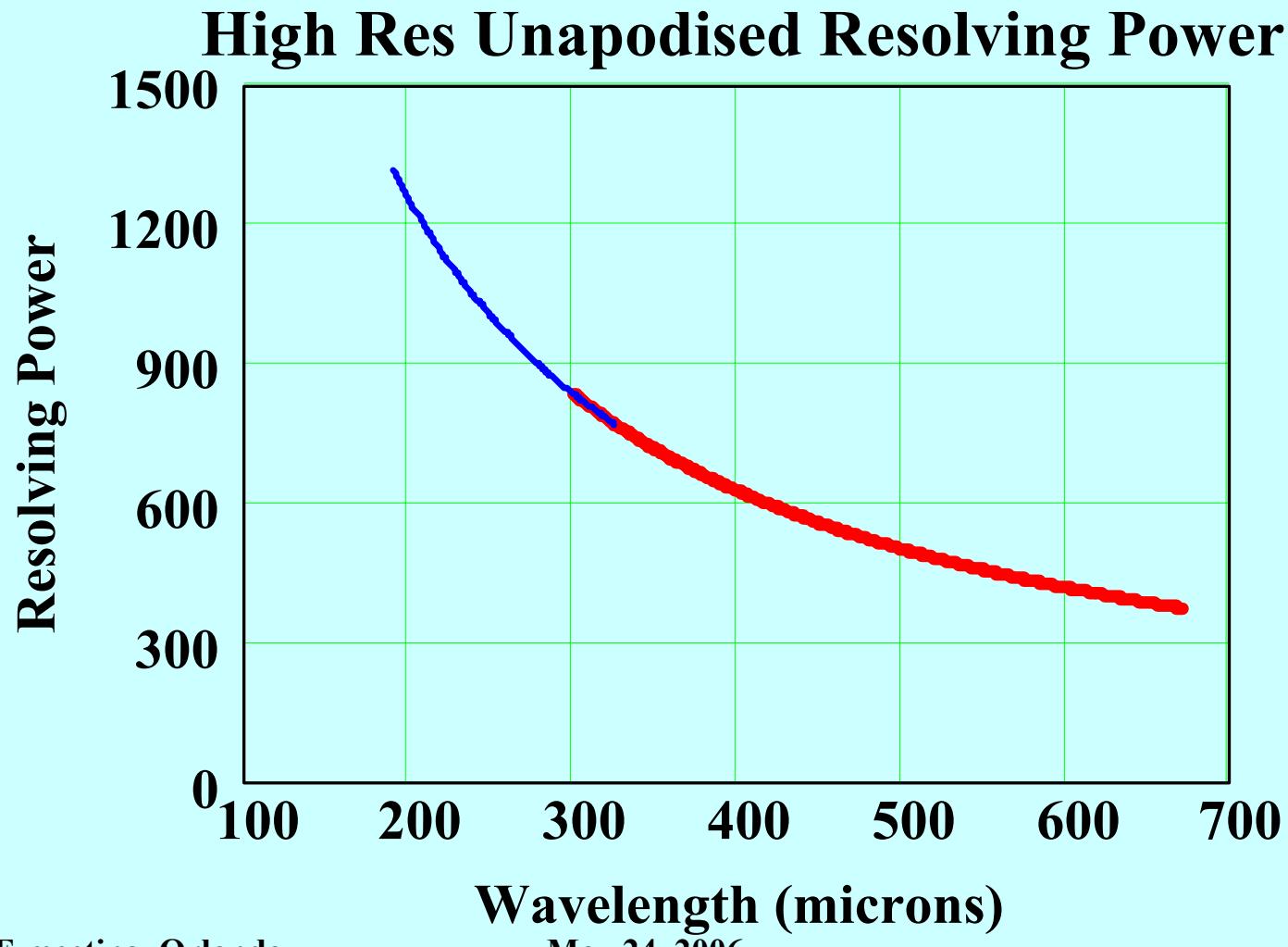


Jiggle  
map

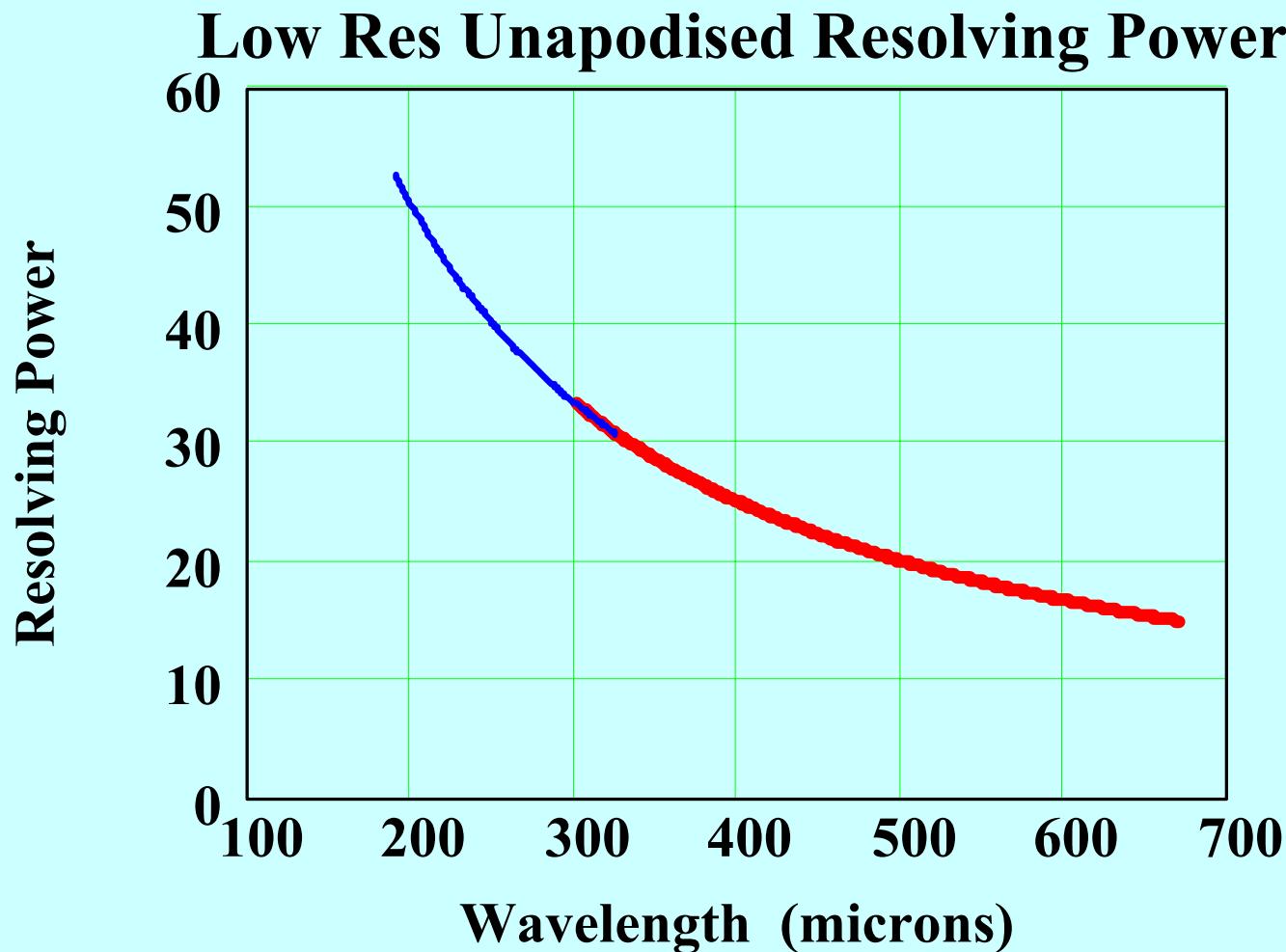
Scan  
map

a  
Point source  
(7-point)/  
sparse map

# FTS Spectral Line Resolving Power ( $\Delta\sigma = 0.04 \text{ cm}^{-1}$ )

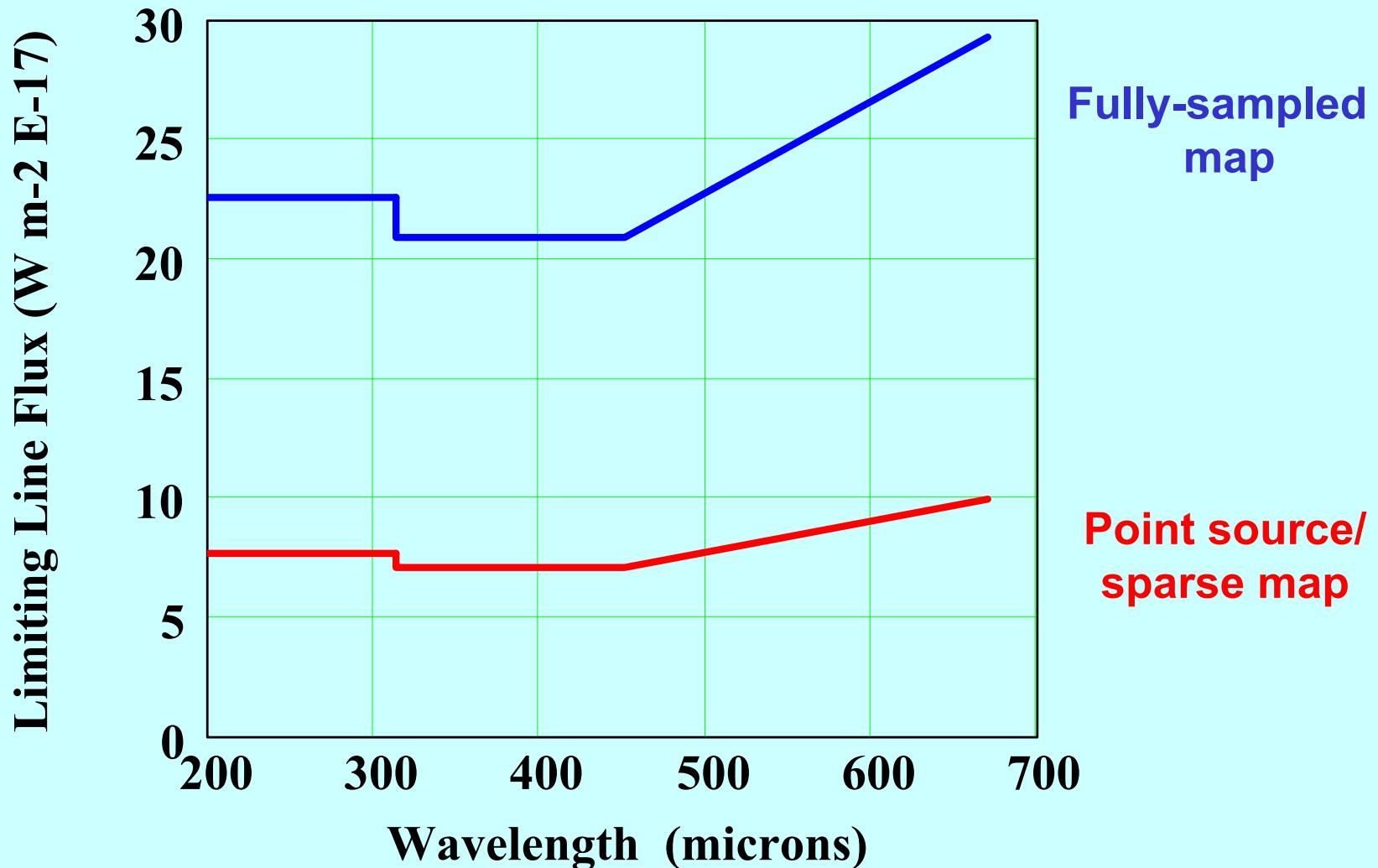


# FTS Spectrophotometry Resolving Power ( $\Delta\sigma = 1 \text{ cm}^{-1}$ )



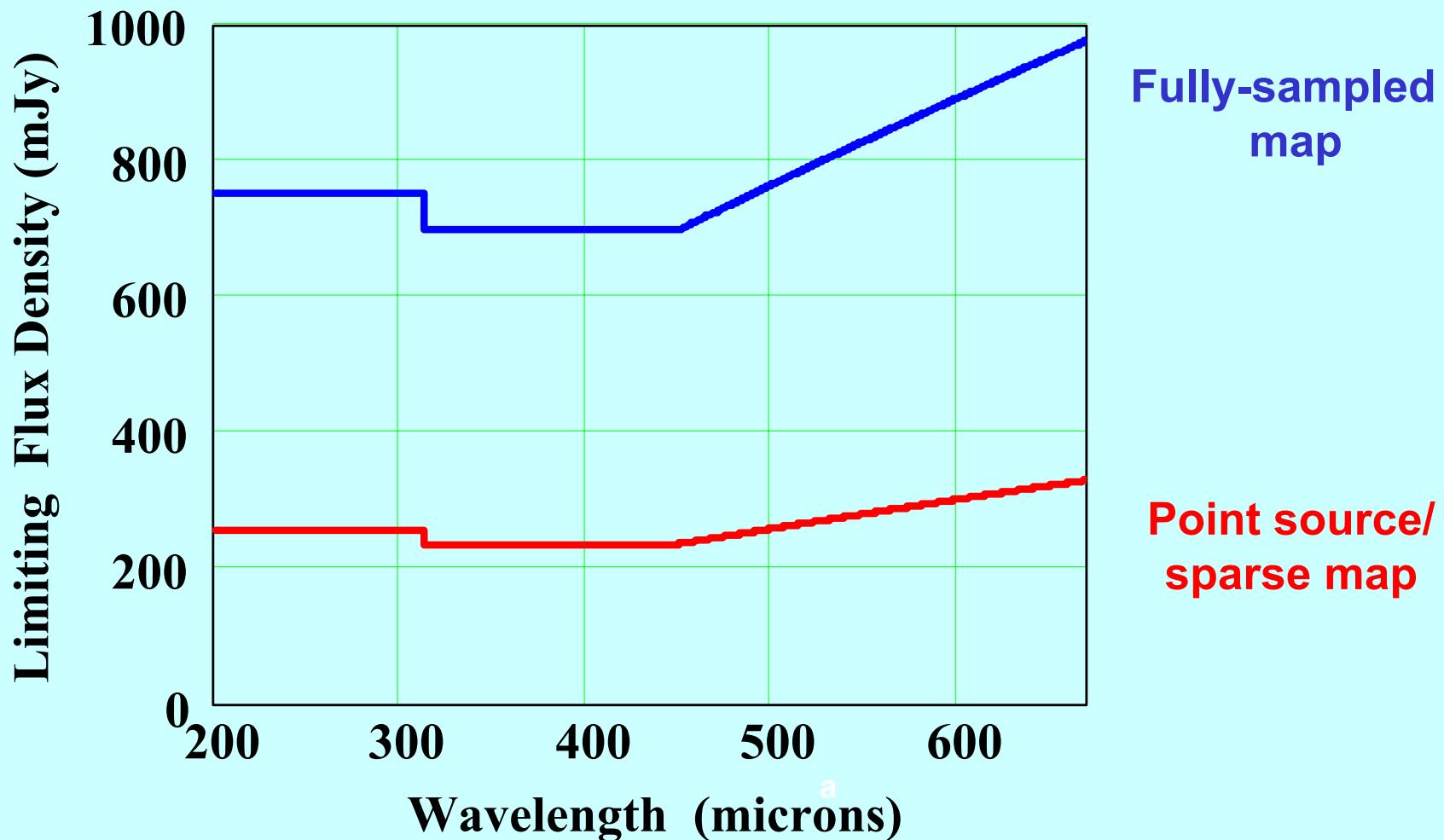
# Sensitivity: Line Spectroscopy

( $5\sigma$  ; 1 hr)



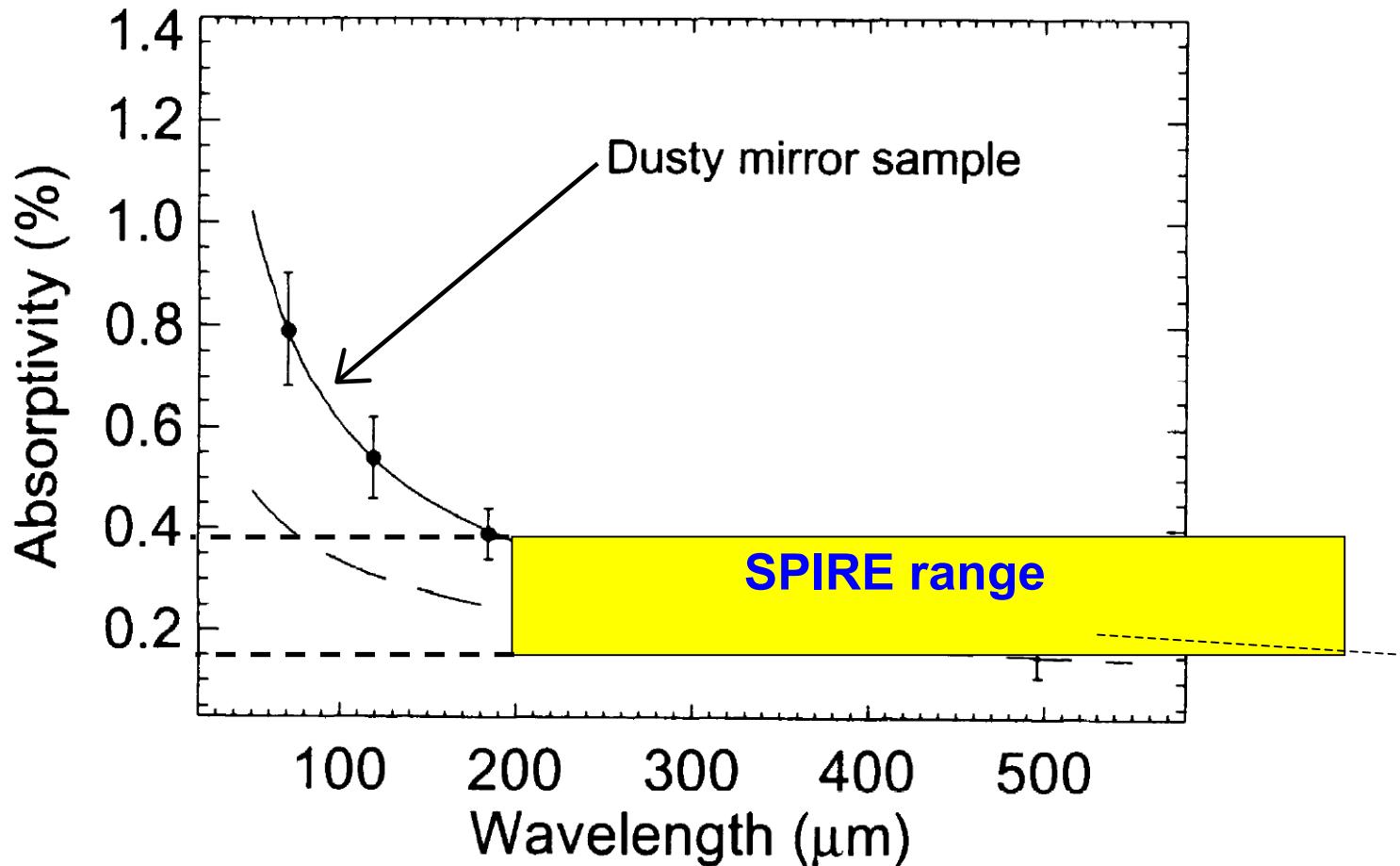
# Sensitivity: Spectrophotometry

( $\Delta\sigma = 1 \text{ cm}^{-1}$ ) (5  $\sigma$  ; 1 hr)

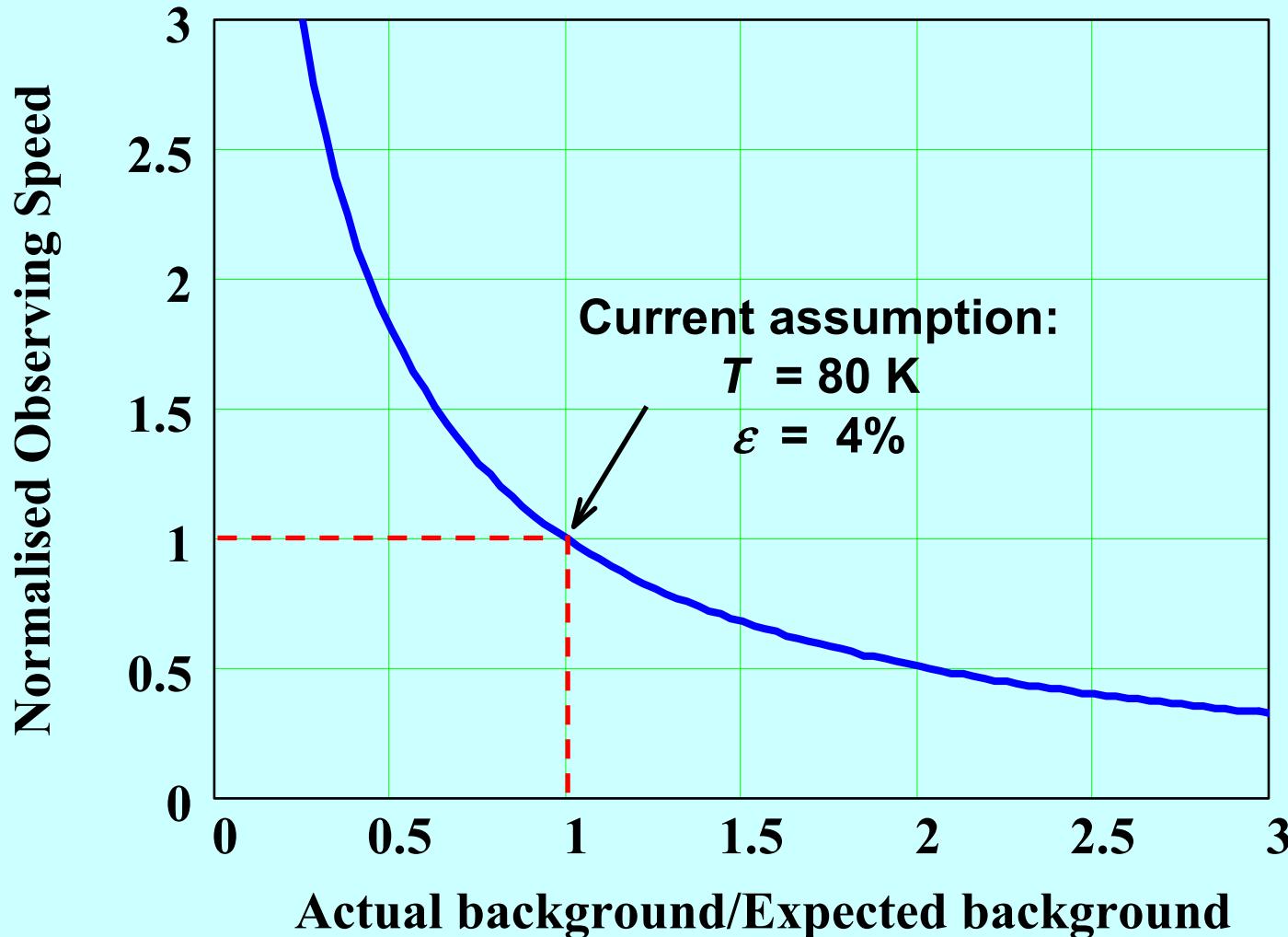


# Herschel Reflector Emissivity

(Fischer et al., Applied Optics, 43, 3765, 2004)

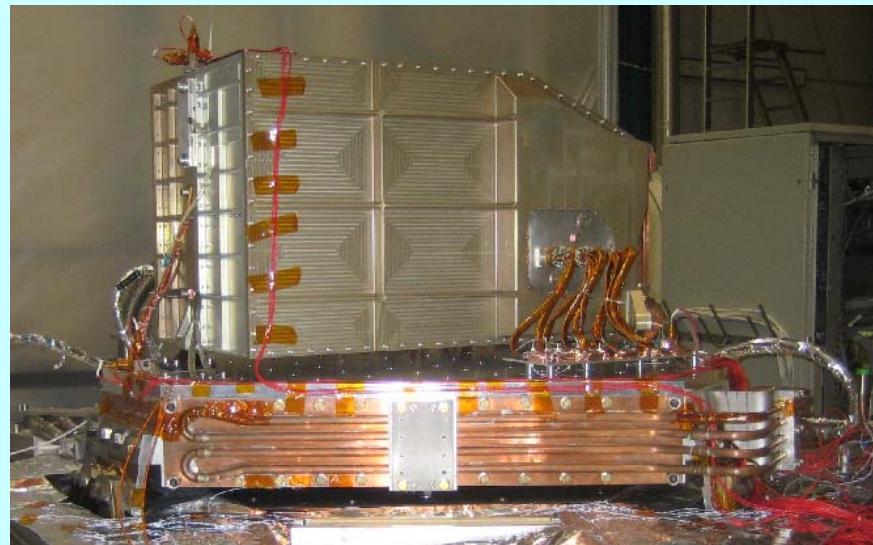


# Observing Speed vs. Telescope Background Power (360 $\mu$ m)



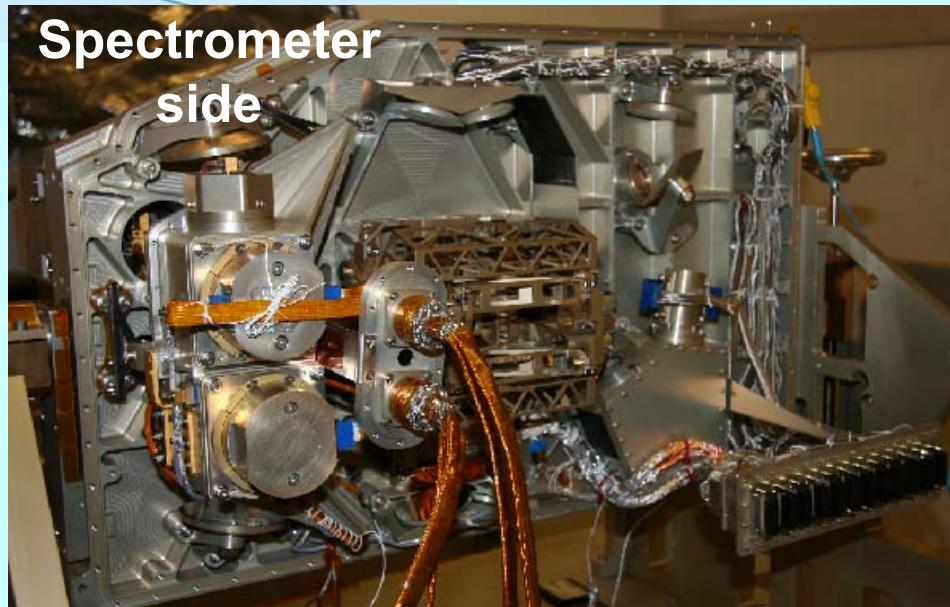
# Current Status

- Flight instrument is being built and tested in stages at Rutherford Appleton Laboratory, UK
- Test campaigns so far
  - 1. Spectrometer side (early 2005)
  - 2. Full instrument (Autumn 2005)
  - Qualification vibration (end 2005)
  - 3. Post-vibration cooldown  
(currently in progress)
- Future steps
  - Installation of flight spectrometer mechanism
  - Final cold vibration
  - Extended instrument evaluation, calibration, observing mode testing
- Integration and test of flight spare
  - Use as a test-bed and for training after flight instrument delivery



# SPIRE Flight Instrument

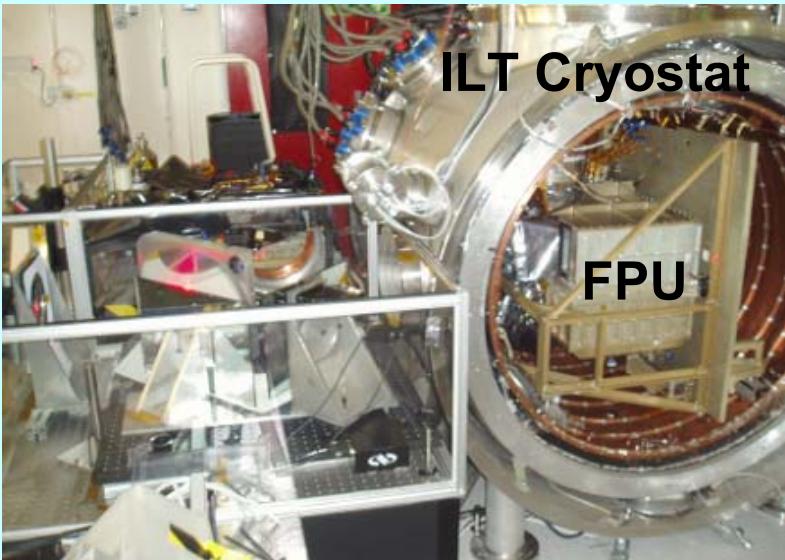
Spectrometer  
side



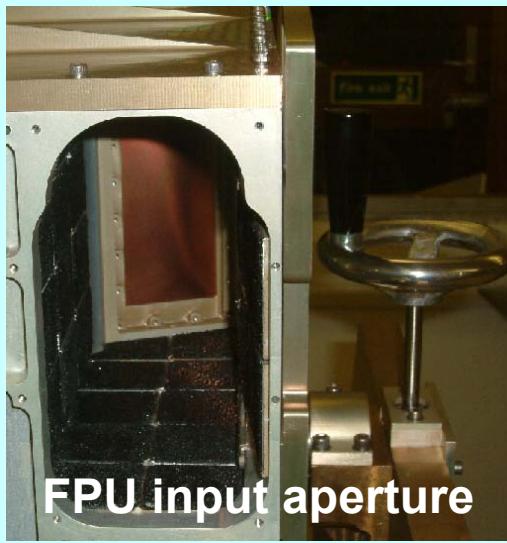
Photometer  
side



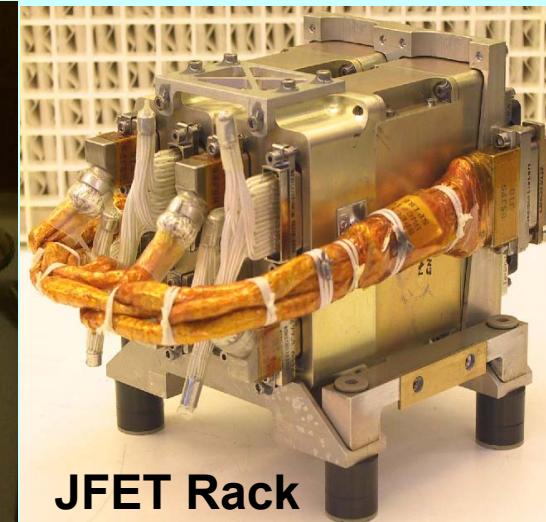
ILT Cryostat



FPU input aperture



JFET Rack





## Other SPIRE-Related Talks and Posters at this Meeting

<b><i>Swinyard et al.</i></b>	<b>6265-12</b>	<b>Optical performance</b>
<b><i>Lim et al.</i></b>	<b>6265-13</b>	<b>Flight instrument testing overview</b>
<b><i>Lindner et al.</i></b>	<b>6265-106</b>	<b>FTS simulator</b>
<b><i>Spencer et al.</i></b>	<b>6265-107</b>	
<b><i>Naylor et al.</i></b>	<b>6265-108</b>	<b>FTS system performance</b>
<b><i>Sibthorpe et al.</i></b>	<b>6270-47</b>	<b>Photometer observing modes</b>
<b><i>Hargrave et al.</i></b>	<b>6275-40</b>	<b>300-mK thermal system</b>
<b><i>Hargrave et al.</i></b>	<b>6275-41</b>	<b>Internal calibrators</b>
<b><i>Nguyen</i></b>	<b>6275-45</b>	<b>Bolometer array performance</b>